

The PROCESS Experiment: Amino and Carboxylic Acids Under Mars-Like Surface UV Radiation Conditions in Low-Earth Orbit

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Abstract

The search for organic molecules at the surface of Mars is a top priority of the next Mars exploration space missions: Mars Science Laboratory (NASA) and ExoMars (ESA). The detection of organic matter could provide information about the presence of a prebiotic chemistry or even biological activity on this planet. Therefore, a key step in interpretation of future data collected by these missions is to understand the preservation of organic matter in the martian environment. Several laboratory experiments have been devoted to quantifying and qualifying the evolution of organic molecules under simulated environmental conditions of Mars. However, these laboratory simulations are limited, and one major constraint is the reproduction of the UV spectrum that reaches the surface of Mars. As part of the PROCESS experiment of the European EXPOSE-E mission on board the International Space Station, a study was performed on the photodegradation of organics under filtered extraterrestrial solar electromagnetic radiation that mimics Mars-like surface UV radiation conditions. Glycine, serine, phthalic acid, phthalic acid in the presence of a mineral phase, and mellitic acid were exposed to these conditions for 1.5 years, and their evolution was determined by Fourier transform infrared spectroscopy after their retrieval. The results were compared with data from laboratory experiments. A 1.5-year exposure to Mars-like surface UV radiation conditions in space resulted in complete degradation of the organic compounds. Half-lives between 50 and 150 h for martian surface conditions were calculated from both laboratory and low-Earth orbit experiments. The results highlight that none of those organics are stable under low-Earth orbit solar UV radiation conditions. Key Words: Mars—Astrobiology—Organic matter—Low-Earth orbit—UV radiation—EXPOSE. *Astrobiology* 12, 436–444.

1. Introduction

SINCE THE FIRST orbital observations of the surface of Mars by the Mariner 4 probe, evidence of the past presence of large expanses of liquid water at the surface of the planet have been discovered (Mangold *et al.*, 2004; Squyres *et al.*, 2004; Poulet *et al.*, 2005; Carter *et al.*, 2010). Some observations suggest that liquid water (a supposed key condition for prebiotic chemistry and the emergence of life) was stable and widespread at the surface of Mars during the first 500 million years of the planet's history (Bibring *et al.*, 2006). As a consequence, Mars is today an object of primary interest for astrobiology in that indications of a prebiotic chemistry, or even biological activity, could be found on the planet. Moreover,

about half the surface of the planet is older than 3.8 billion years because of the absence of global plate tectonics. Therefore, past mineral or organic records of prebiotic chemistry or biological activity could be preserved, and their detection is one of the major objectives of the current Mars Exploration Program (Arvidson, 2000; Morrison, 2001; Hoehler and Westall, 2010).

The search for organic molecules at the surface of Mars started in the 1970s. The *in situ* Viking exploration mission reached the surface of the planet with instruments devoted to the detection of extant biological activity and the search for organic matter in regolith samples. However, no organic matter and no biological activity were unambiguously detected with the Viking instrumentation (Biemann, 1979). At

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that time, the nondetection of organic molecules was considered surprising, even in light of the fact that it is possible that endogenous organic compounds present on Mars for millions of years are not detectable due to burial and degradation processes. Indeed, the interplanetary medium is still an active exogenous source of organic matter (Cottin *et al.*, 1999; Botta and Bada, 2002; Elsila *et al.*, 2009; Schmitt-Kopplin *et al.*, 2010), and it follows that this exogenous organic matter should have been detected by the Viking instruments. Today, two main hypotheses attempt to explain the Viking results: (i) the gas chromatograph–mass spectrometer (GC-MS) instrument would not have been ideally suited to detect low levels of organics in the samples collected by Viking (Navarro-González *et al.*, 2006, 2010) and (ii) organics were effectively absent from the samples collected because of chemical processes that occur at the surface of the planet and lead to the degradation of these molecules (Stoker and Bullock, 1997; Benner *et al.*, 2000; Ten Kate *et al.*, 2006; Stalport *et al.*, 2009, 2010a). Furthermore, it is also important to note that the Viking landers sampled martian soils from two very localized locations. After 40 years, debate with regard to results of the Viking mission is ongoing. The search for organic matter at the surface of Mars, however, began anew with the Mars Science Laboratory 2011 (NASA) and the ExoMars 2018 (ESA) missions. On board these two space probes instruments such as Sample Analysis at Mars (SAM) have the capacity to detect very low levels of organics (Cabané *et al.*, 2004). Also, a new set of laboratory studies, developed within the Mars Organic Molecules Irradiation and Evolution (MOMIE) program, that support these ambitious space programs is devoted to evaluation of the behavior of organic compounds under simulated martian surface environmental conditions (Stalport *et al.*, 2008, 2009, 2010a). The major goals of these studies are (i) the identification of the nature of organic molecules that could be stable or metastable at the surface of Mars, (ii) the assessment of the ability to detect them by *in situ* instrumentation, and (iii) the quantification of their abundance at the surface of the planet.

An assessment of the current martian surface environmental conditions reveals that several processes could be involved in the evolution of organic matter. Among these processes, UV radiation can be particularly efficient. Numerical models predict that the surface of Mars is exposed to an energetic UV flux in the 190–400 nm range (Kuhn and Atreya, 1979; Cockell *et al.*, 2000; Patel *et al.*, 2002). Laboratory experiments have been developed to mimic such radiation conditions and evaluate their impact on organic materials likely to be present at the martian surface (Stoker and Bullock, 1997; Gontareva, 2005; Ten Kate *et al.*, 2005, 2006, Stalport *et al.*, 2008, 2009; Shkrob *et al.*, 2010). Those studies highlighted the degradation of most of the organic molecules. However, it is currently impossible to faithfully simulate the solar UV spectrum and flux in standard laboratory conditions. A good reproducibility of this parameter is very important in order to estimate the kinetic constants (lifetime, photolysis rate constant) for each organic compound. These kinetic constants are themselves essential to model the concentration of organic matter on Mars as a function of the depth, latitude, longitude, geological period, and so on. Therefore, we developed complementary experiments to support laboratory experiments and allow the exposure of samples to the solar spectrum in low-Earth orbit (LEO). This work has been initiated with the UVolution experiment. This experiment

was contained in an ESA Biopan facility outside a Foton automated capsule in 2007 (Guan *et al.*, 2010; Stalport *et al.*, 2010a, 2010b). Another similar LEO study that is the topic of this paper, the PROCESS experiment, was part of ESA's EXPOSE-E mission and was installed outside the European Columbus module of the International Space Station (ISS) from February 2008 to August 2009. This experiment was led by the Laboratoire Interuniversitaire des Systèmes Atmosphériques; the Laboratoire Atmosphères, Milieux, Observations Spatiales; and the Centre de Biophysique Moléculaire; and it was supported by the European and French space agencies (ESA and CNES). The PROCESS samples consist of a set of organic compounds related to cometary or martian environments. This article presents the impact of UV radiation on these organic molecules.

2. Materials and Methods

The PROCESS experiment was part of the science payload of the European EXPOSE-E exposure facility (Fig. 1), which was attached to the outside balcony of ESA's Columbus module of the ISS for 1.5 years (Rabbow *et al.*, 2012). A detailed description of the whole PROCESS experiment and its accommodation in the EXPOSE-E facility is given by Cottin *et al.* (2012).

2.1. Test molecules

In this “Mars” portion of the PROCESS experiment, the stability of the amino acids glycine and serine and the carboxylic acids phthalic acid and mellitic acid were tested in space under Mars-like surface UV radiation conditions. Glycine is one of the most abundant amino acids detected in meteorites and comets (Botta and Bada, 2002; Elsila *et al.*, 2009), and its evolution under simulated martian conditions has already been studied in several laboratories (Ten Kate *et al.*, 2005, 2006; Stalport *et al.*, 2008). The occurrence of serine in meteorites is still uncertain, as it has not yet been detected above contamination levels (Brinton *et al.*, 1998). It is expected to be present on Mars, however.

Phthalic and mellitic acids are aromatic carboxylic acids that have been proposed to be present at the surface of Mars (Benner *et al.*, 2000). Laboratory studies have shown that the evolution of mellitic acid produces a radiotolerant organic compound identified as benzenehexacarboxylic acid–trianhydride (C₁₂O₉) (Stalport *et al.*, 2009). To study a possible photocatalytic interaction of phthalic acid with martian regolith (Shkrob *et al.*, 2010), this acid was also exposed together with JSC Mars-1, a mineralogical analogue of martian soil that is composed of volcanic ashes collected in Hawaii and exhibits similarities to some bright areas of Mars (Allen *et al.*, 1998).

2.2. Flight experiment hardware

For this “Mars” portion of the experiment PROCESS, closed exposure cells (12 mm diameter, 9.1 mm high) (Fig. 2) were used as described in Cottin *et al.* (2012). In brief, each cell was composed of two aluminum cylinders, which were placed in sample carriers that were situated one above the other. Samples in the upper carrier are exposed to UV radiation. Samples in the lower carrier are not exposed and function as dark controls. Each sample cylinder was covered

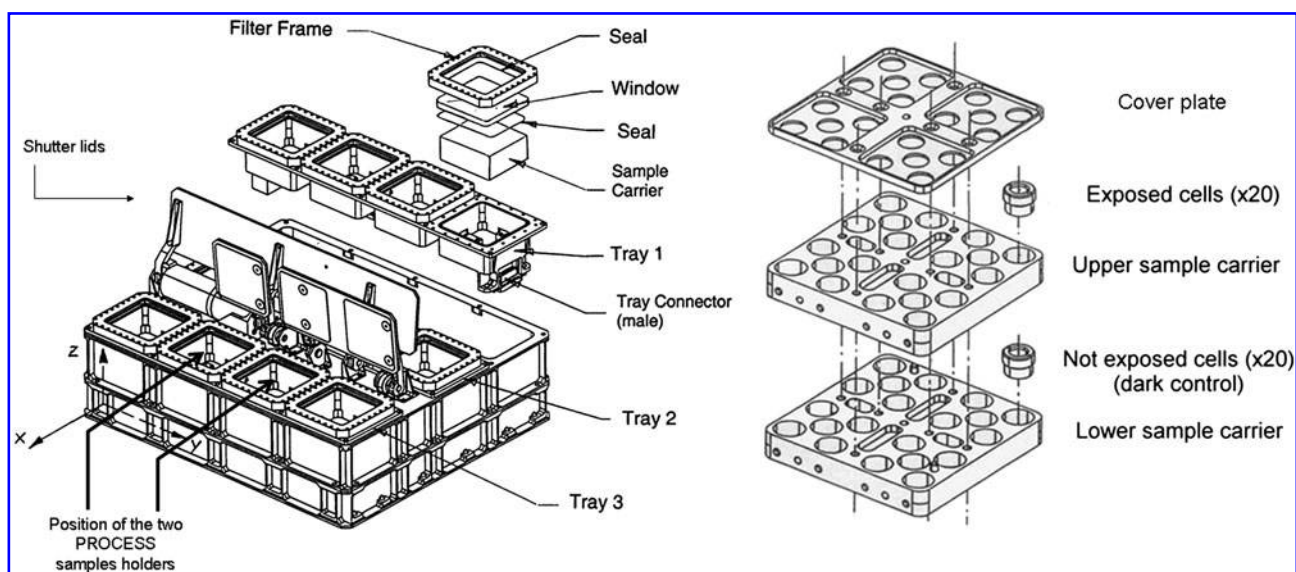


FIG. 1. Left: The EXPOSE facility ($480 \times 520 \times 327.5$ mm) is made of three experiment trays into which four square sample carriers ($77 \times 77 \times 26$ mm) are fitted (left). For PROCESS, two sample carriers were designed to receive 20 Sun-exposed cells and 20 not-exposed cells acting as flight dark control. Right: one of the 2 sample carriers. Pictures courtesy of Kayser-Threde GmbH.

by a quartz window at both ends. After the top cylinder was loaded with one sample on the inner side of the quartz window, the two cylinders were screwed together in an argon atmosphere (1.5 bar). To minimize leaks of the closed exposure cell, it was sealed with a Viton O-ring and an external epoxy structural adhesive layer on the screw thread (2216 B/A, Scotch-Weld). The quartz windows (fused silica) were chosen because this material is transparent in the 190–400 nm wavelength range and efficiently absorbs photons below 170 nm. The spectrum of the UV photons transmitted throughout such a window is close to that which reaches the surface of Mars (190–400 nm wavelength range) (Kuhn and Atreya, 1979; Cockell *et al.*, 2000; Patel *et al.*, 2002). On the EXPOSE-E facility, each cell in the upper carrier that was exposed to solar UV radiation was associated with another

cell directly below it in the lower carrier that contained exactly the same content and was used as a flight dark control (Fig. 1). Some of the quartz windows delivered to us had optical defects, and preparation time for delivery to ESA did not allow for replacement of them. Those identified as defective were placed in the lower, unexposed carrier layers to minimize their impact on our results.

2.3. Loading of the exposure cells

The organic molecules were deposited as a homogeneous thin film on the quartz window (1 mm thick, 9 mm diameter) of each top cylinder by using a vacuum sublimation chamber (Guan *et al.*, 2010). The films were a few hundred nanometers thick as measured by interferometry (Cottin *et al.*, 2012). For

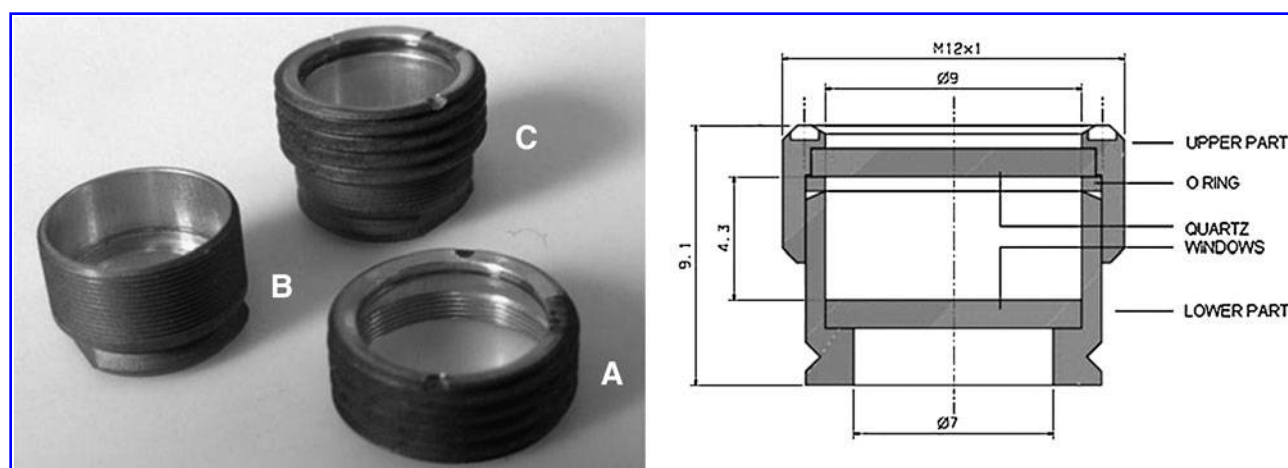


FIG. 2. Design of a closed cell. Two aluminum cylinders are screwed into each other: (A) Top; (B) bottom; (C) complete cell composed of a top (A) and a bottom (B). Two quartz windows are glued at both ends to allow the analysis of molecules inside the cell by spectroscopy. Sealing (relative to lab atmosphere or vacuum in space) is ensured by a Viton O-ring. Right picture courtesy of COMAT Aerospace.

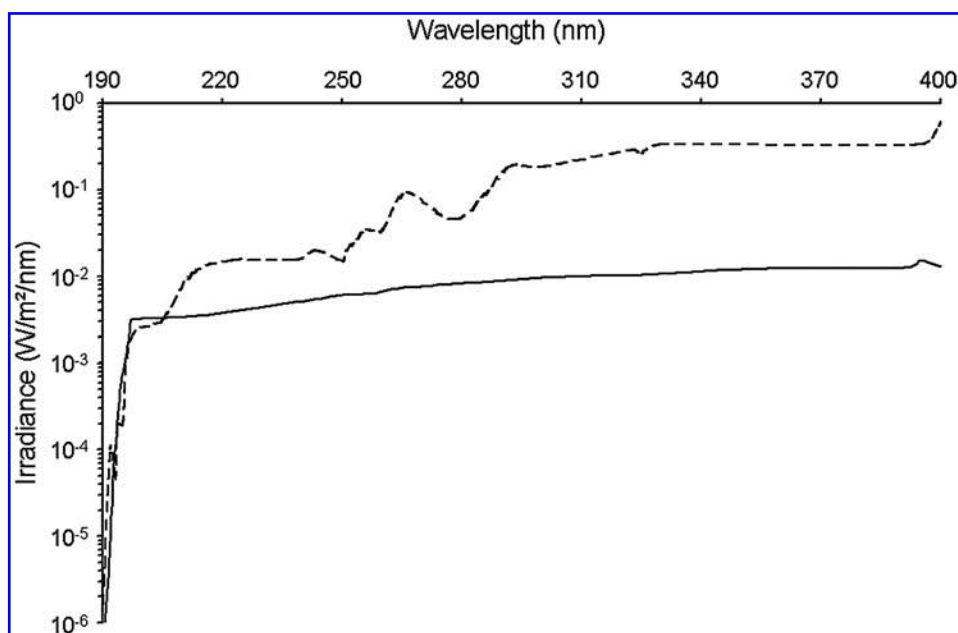


FIG. 3. UV spectrum of the xenon lamp (black line) compared to the solar UV spectrum on the martian surface predicted by Patel *et al.* (2002) at equatorial noontime (dotted line).

the phthalic acid-mineral combination, a stepwise approach was used. First, a mineral film was deposited on the quartz window as follows: the JSC Mars-1 soil was sieved, and the $<40 \mu\text{m}$ fraction was collected and dispersed in ethanol; the suspension was deposited on the quartz window and then heated to 50°C to evaporate the ethanol. The mineral film was several hundred micrometers thick as measured by interferometry. The organic film was then deposited on the surface of the mineral phase by using the sublimation chamber. For each organic compound or compound-mineral combination, eight parallel samples (*i.e.*, eight cells) were produced.

2.4. Test parameters

The eight samples were split into four pairs:

- (i) Two “flight Sun-exposed” samples that were located in the upper sample carrier of the flight experiment (Fig. 1). After 1.5 years in space, these samples received a UV (200–400 nm) radiation fluence of $5.2 \times 10^5 \text{ kJ/m}^2 \pm 15\%$, as calculated from the ISS orbital parameters by the company RedShift (St Niklaas, Belgium) (Rabbow *et al.*, 2012). These samples also experienced other space environmental conditions such as cosmic radiation of a total dose of 295.6 mGy (Berger *et al.*, 2012; Dachev *et al.*, 2012) and temperature fluctuations between -21°C and $+61^\circ\text{C}$ (Rabbow *et al.*, 2012).
- (ii) Two “flight dark control” samples that were located in the lower sample carrier of the flight experiment (Fig. 1). These samples experienced the same environmental conditions as those that were flight Sun-exposed but no insolation.
- (iii) Two “ground thermal cycling” samples that were kept in the Planetary and Space Simulation Facility at the German Aerospace Center (DLR), Cologne, Germany, under low pressure (1.7×10^{-8} bar) and submitted to thermal cycles, which reproduced the recorded flight temperature (Rabbow *et al.*, 2012).

- (iv) Two “ground, constant temperature” samples that were kept at low pressure (1.7×10^{-8} bar) and constant temperature (5°C) at the DLR facilities for the entire duration of the mission.

2.5. Mars Organic Molecules Irradiation and Evolution facility

To study in parallel the kinetics of evolution of organics under simulated martian conditions, a Mars simulation setup called the Mars Organic Molecules Irradiation and Evolution (MOMIE) facility was developed as described by Stalport *et al.* (2009) to reproduce the UV radiation climate at the martian surface. A UV source (Xenon arc lamp, LOT-ORIEL) was used that released photons in the 190–400 nm wavelength range (Fig. 3). For this study, the organic molecules were deposited by vacuum sublimation on MgF_2 windows with transmission properties in the IR down to 1000 cm^{-1} .

2.6. Sample analyses

The samples were analyzed before and after the mission by Fourier transform infrared (FTIR) spectroscopy (Perkin Elmer BXII spectrometer, resolution 4 cm^{-1} , eight scans, spectral domain $4000\text{--}2000 \text{ cm}^{-1}$) by direct transmission through the quartz window. The IR areas of the spectral feature’s peaks before and after the mission were compared to measure the photolysis efficiency.

In the laboratory studies in which MOMIE was used, IR spectra were measured with the same instrument through the MgF_2 windows (spectral domain $4000\text{--}1000 \text{ cm}^{-1}$) as a function of photolysis time, and the evolution of the photo-processing was parameterized according to

$$\text{Ln} \frac{[A]_t}{[A]_0} = -J \cdot t \quad (1)$$

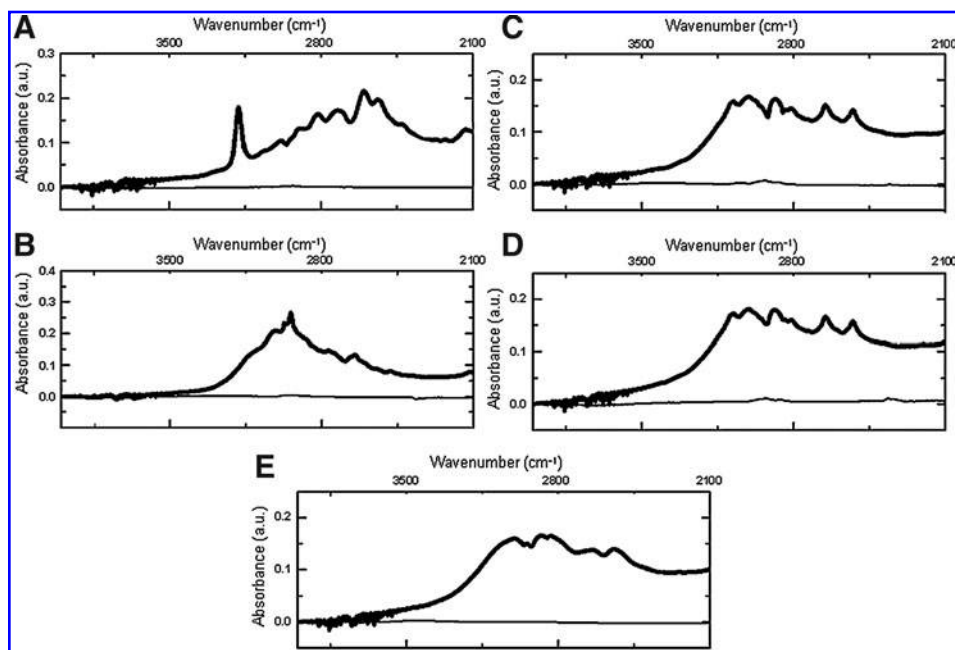


FIG. 4. IR spectra (in the 4000–2000 cm^{-1} wavelength range with a 4 cm^{-1} resolution) of a “flight Sun-exposed” sample before (bold) and after (fine) spaceflight; glycine (A); serine (B); phthalic acid (C); phthalic acid with JSC Mars-1 (D); mellitic acid (E) (a.u. = arbitrary units).

with $[A]$ the abundance A of the molecule, t the irradiation time, and J the photolysis rate ($[\text{time}]^{-1}$).

To study the gas phase after the mission, gas chromatography coupled to mass spectrometry (GC Varian CP 3800 and MS Varian Saturn 2200) was applied in addition to FTIR spectroscopy. First, each cell was immersed in dichloromethane (CH_2Cl_2) for 75 min to degrade the epoxy adhesive. The samples were then opened in a specific airtight device called an “analytic cell” (Cottin *et al.*, 2008, 2012). This analytical cell is interfaced to the GC-MS. Once a closed cell is opened inside the analytical cell, the gaseous samples are transferred to the GC injection loop and injected at 175°C in the splitless mode to maximize the quantity of gaseous species analyzed. The GC column was a Restek Q-plot capillary column ($30 \text{ m} \times 0.25 \text{ mm} \times 10 \mu\text{m}$). This column is efficient for light hydrocarbons (C1–C12). The initial temperature of the column was 40°C . After 4 min at 40°C , the GC oven was ramped to 150°C at a rate of $10^\circ\text{C}/\text{min}$. The final temperature of the column was maintained at 150°C for 25 minutes. Pressure was set from 15 to 25 psi at 1.5 psi/min. Products were measured with a quadrupole MS (14–200 m/z).

3. Results

3.1. Solid phase: Fourier transform infrared analysis by transmission

Fourier transform infrared spectra of the different test samples were taken at an interval of 4 years, either before the flight in July 2007 or after the mission and retrieval in July 2011. In Fig. 4, the IR spectra of the “flight Sun-exposed” samples are presented. After 4 years with 1.5 years in space, the spectral features of the initial sample had disappeared in all “flight Sun-exposed” samples, which indicates 0% recovery of the Sun-exposed samples after spaceflight (with the exception of one “flight Sun-exposed” glycine sample that showed 100% recovery, which means that this cell was not properly exposed to the Sun). In contrast, the percent re-

covery of most samples that were not directly exposed to extraterrestrial solar UV radiation (“flight dark control,” “ground thermal cycling,” and “ground constant temperature”) was 100% (data not shown). Exceptions were one “ground thermal cycling” glycine sample with 83% recovery and one “flight dark control” phthalic acid sample with 92% recovery. Both “flight dark control” samples of phthalic acid with JSC Mars-1 showed a reduced recovery (93% and 74%) as well as one “ground thermal cycling” sample of this organic-mineral combination (66%). By comparing the results of the “flight Sun-exposed” samples with those of the “flight dark control” samples and both ground controls, the 0% recovery of the “flight Sun-exposed” samples can only be interpreted by photodegradation of these samples. The only interpretation of these results is that a UV (200–400 nm) radiation fluence of $5.2 \times 10^5 \text{ kJ}/\text{m}^2 \pm 15\%$ completely photolyzed all test samples.

The laboratory simulations by use of MOMIE provided supportive information about the evolution kinetics of the organic molecules used in this study. The photolysis of the four organics investigated follows a first-order decay, Eq. 1 (Fig. 5). From these curves, the half-lives of glycine, serine, phthalic acid, and mellitic acid under Mars-like UV radiation were determined (Table 1).

3.2. Gas phase: gas chromatography–mass spectrometry analysis

The gaseous components of the flight cells and the ground control cells were analyzed by GC-MS after exposure, retrieval, and completion of the IR spectroscopy. For all cells, in addition to the expected argon, the gases N_2 and CH_2Cl_2 were detected. In the gas phase of the “flight Sun-exposed” cells, the following hydrocarbons were measured: ethylene, ethane, propane, isobutane, and butane. These hydrocarbons were not found in the other cells, the “flight dark control,” or the “ground control” cells, except trace amounts of ethane

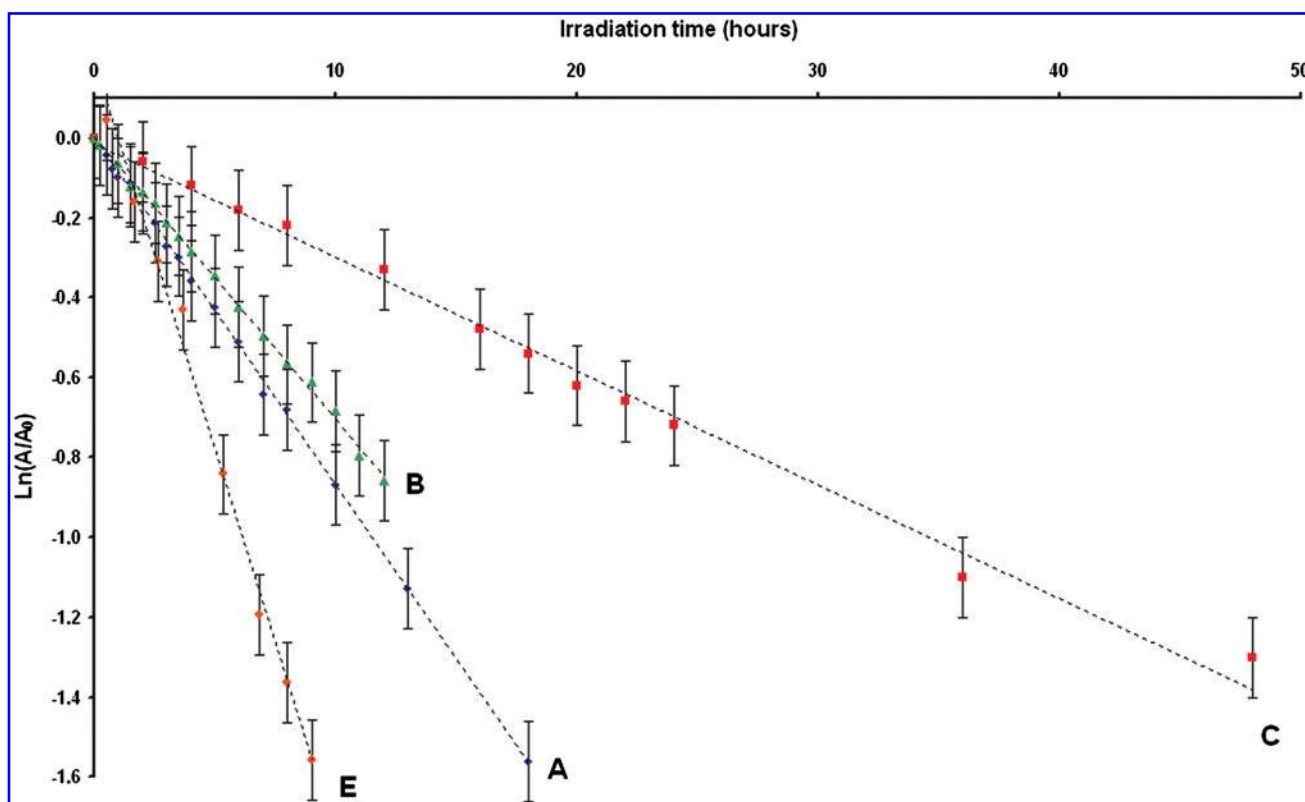


FIG. 5. Results of the photolysis rate of the organic targets from laboratory simulations (MOMIE). The natural logarithm of the normalized integrated absorbance ($\text{Ln}(A/A_0)$) is plotted against time for glycine (A), serine (B), phthalic acid (C), mellitic acid (E). The linear correlation for mellitic acid has been obtained from one band. Color images available online at www.liebertonline.com/ast

that were detected in one “flight dark control” cell of glycine and one of serine.

The presence of N_2 is the consequence of a leak during the injection of the samples into the GC-MS. The presence of CH_2Cl_2 (the solvent used to eliminate the epoxy adhesive) could be explained by its accumulation during the dissolution of the adhesive. The presence of CO_2 could be explained by a leak during the injection into the GC-MS, but it could as well be a product of photodegradation.

Results are somewhat puzzling because the same hydrocarbons were detected from cells containing different organic molecules, and the detected molecules were not typical photodegradation products of amino acids. Their gaseous products should have been mostly CO_2 and HCN (Ehrenfreund *et al.*, 2001) and not the hydrocarbons detected in this

study. An explanation for the appearance of those hydrocarbons could be the photodegradation of the exposure cells themselves. Laboratory simulation tests conducted with the MOMIE experiment on the different cell components (coating, Viton O-rings) showed that UV radiation of an empty cell filled with O-rings produced the same gaseous compounds as measured in the “flight Sun-exposed” cells. Additional products were pentane, cyclopentane, cyclohexane, hexane, benzene, and toluene. Quantitative aspects cannot be addressed in the AMINO closed cell since they suffer from non-negligible leaking over such a long-duration experiment: from 5% to 100% between the filling and the postflight analysis. As the evolution of the leaking over time is not known, the amount of gaseous photoproducts cannot be precisely estimated from the gas chromatography measurements. Therefore, it is unknown whether the products detected in the gas phase of the “flight Sun-exposed” cells came from the degradation of the sample cells themselves or from a combination of such a contamination and actual photoproducts of the exposed organic samples. For future space experiments, a new generation of extra-tight cells has been developed, which should overcome these limitations.

TABLE 1. HALF-LIVES OF GLYCINE, SERINE, AND PHTHALIC ACID DETERMINED FROM MOMIE EXPERIMENTS AND CALCULATED FOR THE SAME RANGES OF UV RADIATION (200–400 NM) OCCURRING IN LEO AND ON THE SURFACE OF MARS

Molecule	Half-life MOMIE (h)	Calculated half-life in LEO (h)	Calculated half-life on the surface of Mars (h)
Glycine	7.9 ± 0.2	17.3 ± 0.5	51.3 ± 1.3
Serine	11.6 ± 1.7	25.4 ± 3.7	73.9 ± 10.8
Phthalic acid	23.3 ± 0.6	51.0 ± 1.3	148.3 ± 3.8

4. Discussion

It was the aim of this study to determine the evolution of the amino acids glycine and serine and the carboxylic acids phthalic acid and mellitic acid under Mars-like surface UV radiation conditions with the filtered extraterrestrial solar electromagnetic spectrum. The FTIR results show that the

direct exposure to simulated martian UV radiation for 1.5 years led to the total photodegradation of these organics. The presence of a mineral phase (phthalic acid with JSC Mars-1) did not change the results.

Due to the complete photodegradation of the exposed samples, a quantitative discussion about their photolysis kinetics is limited to the estimation of half-life upper limits and comparison with Mars simulation studies conducted in the laboratory (MOMIE). The integrated flux in the wavelength region 200–260 nm is for the Sun 3.2 W/m^2 at 1 AU (Thuillier *et al.*, 2004), whereas it is $7 \pm 2 \text{ W/m}^2$ for the solar simulator of MOMIE. The half-lives calculated from our laboratory experiments (Fig. 5) and rescaled to LEO conditions are $(17.3 \pm 0.5) \text{ h}$, $(25.4 \pm 3.7) \text{ h}$, and $(51.0 \pm 1.3) \text{ h}$ for glycine, serine, and phthalic acid, respectively (Table 1). Half-life values for our samples exposed in LEO are $< 220 \text{ h}$ (upper limit considering a $< 1\%$ percent recovery and a first-order decay, Eq. 1). The two sets of data are consistent and show that the laboratory simulations with the MOMIE setup provide credible values of kinetic parameters for the evolution of organic matter.

On Mars, the integrated flux in the 200–260 nm wavelength range has been given as $1.1 \pm 0.2 \text{ W/m}^2$ (Kuhn and Atreya, 1979; Cockell *et al.*, 2000; Patel *et al.*, 2002). Taking the MOMIE laboratory data (Fig. 5) and rescaling them to the martian conditions by cross calculation would give half-lives of glycine, serine, and phthalic acid on Mars of $(51.3 \pm 1.3) \text{ h}$, $(73.9 \pm 10.8) \text{ h}$, and $(148.3 \pm 3.8) \text{ h}$, respectively (Table 1). From these values, it is possible to estimate the amount of these organic compounds in the martian regolith over time. The accumulation of such organic molecules would be due to the difference between their production/intake rates, which have been estimated to be between $2.4 \times 10^7 \text{ g/year}$ (Benner *et al.*, 2000) and $2.4 \times 10^8 \text{ g/year}$ (Flynn, 1996), and the different pathways that may lead to their destruction, for example, by UV radiation. Assuming that the organic compounds were uniformly deposited at the surface of Mars and mixed with the regolith up to a depth of 1 cm [UV radiation is believed to penetrate only the first millimeter of the martian regolith, except in water ice, where the penetration would be several centimeters (Cockell and Raven, 2004)], and assuming only a few centimeters of the soil would be recycled by winds, then the maximal concentration of organics by weight would be less than 1 ppt (with a maximal repartition on the surface of Mars of about $3 \times 10^{-7} \text{ g/m}^2$ and a density of the martian regolith of about 4 g/cm^3) (see also Stalport *et al.*, 2010a). These data will be implemented into a future model focused on the concentration of organic matter at the surface of Mars according to the regolith depth, latitude, longitude, and seasons. The goal of this model will be to predict which compounds at what level could be detected on Mars by an *in situ* mission and what their origin (biotic versus abiotic) might be.

Degradation photoproducts are certainly released in the gas phase (and by extrapolation in the martian atmosphere), but they cannot be securely dissociated from contamination produced by the sample cells themselves. Therefore it is not possible to derive from our analyses reaction mechanisms of the photolysis of our organic samples in the martian environment. Such limitations due to our hardware have been investigated and will be taken into account for future LEO experiments.

5. Conclusion

Organic molecules of astrobiological interest were exposed to space conditions, and especially solar UV photons, during the PROCESS LEO experiment. Some of the samples were selected to study the behavior and stability of organic molecules (glycine, serine, phthalic acid, mellitic acid) that may be present on Mars. Our data show that the targeted molecules are not photostable because they were totally destroyed after the long-duration exposure to solar UV radiation, which was much longer than the duration of the UVolution experiment on board the ESA Biopan facility in 2007 (Stalport *et al.*, 2010a). Laboratory experiments confirmed this result and provided crucial kinetic parameters to understand the preservation rates of organic matter at the surface of Mars.

Complementary experimental conditions, both in the laboratory and in LEO, are essential to understand the evolution of the samples. However, the experimental hardware for LEO experiments will have to be improved to eliminate the main limitations highlighted in this paper: (1) the contamination of the gas phase in the closed cells, (2) the low amount of data (limited amount of samples and replicates), (3) measurements only available before and after the experiment. Future experiments scheduled outside the International Space Station for late 2013 [such as Photochemistry on the Space Station (PSS) on the EXPOSE-R2 facility] will be prepared with a homogeneous and well-characterized new generation of closed cells, which will minimize leaks and contamination of results. Such new cells are currently under development and should be ready for the next experiments in which more space will be allocated such that more samples will be exposed (issue 2). A new next investigative step will be taken to address issue 3 with new generations of exposure instruments, which will include *in situ* diagnostics [O/OREO nanosatellite, OREOCube project for the ISS (Nicholson *et al.*, 2011)] and the VITRINE project by the Centre National d'Études Spatiales.

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Author Disclosure Statement

No competing financial interests exist.

Abbreviations

DLR, Deutsches-Zentrum für Luft- und Raumfahrt e.V. (German Aerospace Center); FTIR, Fourier transform

infrared; GC, gas chromatograph; ISS, International Space Station; LEO, low-Earth orbit; MOMIE, Mars Organic Molecules Irradiation and Evolution; MS, mass spectrometer.

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